

## The Study on Performance of MEMS IMU for Launch Vehicle under high vibration environment

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Inertial Measurement Unit (IMU) installed gyroscopes and accelerometers is the one of the most important component for navigation and guidance system of launch vehicles. The sensors to get the information about acceleration and angle rate of rockets require high accuracy, high reliability, and environment resistance. That is why the IMU for launch vehicle has been large in size and costly.

On the other hand, IMU for commercial use as car navigations, controllers of video games, and smartphones is installed the sensors using Micro Electro Mechanical Systems (MEMS) technology. MEMS devices are produced cheaply in large quantities during the same production process as semiconductor, and have been developed for accuracy improvement.

There are issues related to the accuracy of sensors and the resistance of vibration environment, if the IMU using MEMS sensors, called MEMS IMU, is applied to launch vehicles. For the solution of the problem that the MEMS IMU for launch vehicle require high accuracy, we have developed the MEMS gyroscope called MARS. The high accuracy of MARS is achieved by digital temperature correction, and the bias instability has brought up to 0.1deg/h. [1]

And we have the system-managed approach toward solving the problem of vibration environment. To see whether resist and maintain its performance in the vibration environment of H-IIA and Epsilon rocket, We have produced the IMU installed high accuracy MEMS gyroscopes and accelerometers and measured experimentally. The model numbers are SiIMU02 and HGM-02A. Figure 1 shows the appearance and Table 1 shows the specification of HGM-02A. We have tested MEMS IMU in the vibration environment of H-IIA and Epsilon rocket, using both of the vibration exciter and actual flight of rocket. Although MEMS sensors vibrate itself for sensing mechanism, MEMS IMU has maintained its performance basically.



Fig. 1. MEMS IMU (HGM-02A).

Table 1. Specification of MEMS IMU (HGM-02A).

Item	Specification
Profile	74(W)×74(L) ×60(H)[mm]
Mass	500[g]
Angle Rate Measurement Range	±100deg/s
Angle Rate Measurement Accuracy	1deg/s rms
Angle Rate Measurement Resolution	0.001deg/s
Acceleration Measurement Range	±2G
Acceleration Measurement Accuracy	0.2G rms
Acceleration Measurement Resolution	0.1mG

### References

[1] Takeshi Sasada, Eri Shimane, Hiroshi Nishida, Takafumi Moriguchi, Ryohei Uchino, Futoshi Magosaki, *An Interim Report on the Development of High-Accuracy MEMS Gyros for Space Applications*, Proceedings of 57th Space Sciences and Technology Conference, 2013-10.